

**2009 Leonardo
AW139**

**#31239
F-HHJM**



CONTACT

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VIRTUAL TOUR

2009 Leonardo AW139

#31239 | F-HHJM



TOTAL HOURS

2,164 hours since new

ENGINES

Pratt & Whitney PT6C-67C

#	S/N	Total Time
1	PCE-KB0513	2,164
2	PCE-KB0493	2,164

HIGHLIGHTS

New ‘FiveStar’ VIP interior & paint
@ Leonardo in 2022

6.8T MTOW, Avionics ph7

EAPS, floats, aux tanks, AVCS

Fresh 4 yr inspection @ Leonardo

Hangared Paris, France

AIRFRAME

Aluminium alloy fuselage

Acrylic windshields

Overhead cockpit transparent windows

Lower cockpit transparent windows

Cockpit ram air adjustable outlets

Heated Pitot tubes (2) with Pitot heat
failure warning (2)

Heated static intakes (2)

One cockpit and one passenger cabin fire
extinguisher

Pilot doors (2) with push-out emergency
exit window (with “storm window” on pilot
side)

Plug-in sliding doors (2) for passenger
cabin access with four locking latches

Passenger doors stop in fully open posi-
tion

Six cabin push-out emergency exit
windows (two on each passenger cabin
door)

Baggage compartment with two lockable
baggage compartment doors (LH and RH)

Wheeled retractable tricycle landing gear
(two wheels on nose gear and brakes on
main landing gear)

Structural provisions for nose landing gear
doors

Landing gear mooring

Upper deck mooring provisions

Hoisting and jacking fittings

Tail boom and vertical fin

Horizontal stabilizers with composite
“winglets”

Pilot and co-pilot windshield wipers

Maintenance steps for access to upper
deck on both sides

Upper deck cowlings (one front forward
sliding, two side opening for engine
access)

Steps for cockpit access

ROTORS AND CONTROLS

Fully articulated Main Rotor (M/R) with five composite blades, five
elastomeric bearings and five hydraulic dampers

Rotating M/R flight controls

Main rotor hub beanie

Main dual servo actuators (3)

Fully articulated Tail Rotor (T/R) with four composite blades, four
elastomeric bearings and four elastomeric dampers

Rotating T/R flight controls

One T/R dual servo actuator

Force trim system

Pilot fixed flight controls (cyclic, collective, anti-torque pedals)

Co-pilot fixed flight controls (cyclic, collective, anti-torque pedals)

Dual digital 3-axis linear actuators

Provision for main rotor tracking (magnetic pick-up)

Spec updated: May 13, 2026

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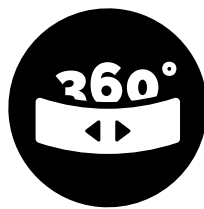
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POWER PLANT AND FUEL SYSTEM

Two Pratt & Whitney Canada PT6C-67C turbo-shaft engines

Two independent FADEC systems with auto-start and engine control functions for normal, emergency and training operations

Two magnetic chip detectors

Two integrated and independent engine oil coolers

Separation firewall protection for each engine

Fire detection system

Fire extinguisher system (2 bottles)

Steel engine exhausts and ejectors

Two independent crashworthy fuel cells

Two supply pumps on engines

Two submerged booster pumps in fuel tank

Two fuel filter assemblies

Two engine back-up controls, mechanical and electrical

Two manual engine start and ignition systems

TRANSMISSION DRIVE SYSTEM AND HYDRAULIC SYSTEM

Main transmission with two direct drive engine inputs

Three main transmission chip detectors/debris collectors with burning capability

Two freewheel units

Four strut rods for upper deck attachment and one anti-torque plate

Intermediate gearbox with sight gauge and magnetic drain plug/chip detector

Tail gearbox with sight gauge and magnetic drain plug/chip detector

Dual independent, redundant hydraulic systems

Two hydraulic Power Control Modules (PCM)

Three main transmission driven hydraulic mechanical pumps for controls, landing gear and wheel brake actuation

One electrical pump for control checks on ground (engines off)

Air conditioning compressor quill

ELECTRICAL SYSTEM

DC primary power generation: 28 V DC regulated voltage provided by two independent 300 A starter generators

DC external power receptacle

Two nickel-cadmium batteries: one 44 Ah (20 cells) main battery plus one 27 Ah auxiliary battery

DC starter generator control units (2)

Navigation lights (3)

Anti-collision light

External emergency landing lights on the sponsons (2)

Retractable/rotating landing/taxing light

Baggage compartment lights (3) and smoke sensor

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STANDARD AVIONIC PACKAGE

1st PFD (Pilot Primary Flight Display) 8" x 10" (external) colour Active Matrix Liquid Crystal Display as part of Honeywell Primus Epic System for visualization of flight and navigation data

2nd PFD (Co-pilot Primary Flight Display) 8" x 10" (external) colour Active Matrix Liquid Crystal Display as part of Honeywell Primus Epic System for visualization of flight and navigation data

1st MFD (Pilot Multi-Function Display) 8" x 10" (external) colour Active Matrix Liquid Crystal Display as part of Honeywell Primus Epic System for visualization of flight and navigation data

2nd MFD (Co-pilot Multi-Function Display) 8" x 10" (external) colour Active Matrix Liquid Crystal Display as part of Honeywell Primus Epic System for visualization of flight and navigation data

One Electronic Standby Instrument System (ESIS) (attitude, airspeed, altitude, vertical speed, compass, and ILS data)

Two Attitude and Heading Reference Systems (AHRS)

Two Air Data Modules (ADM)

Two flux valves

Two Display Controllers (DC)

Two Cursor Control Devices (CCD)

One Reversion Control Panel (RCP)

One Display Dimming Control Panel (DCP)

One Stability Augmentation System (SAS) control panel

Two Remote Instrument Controllers (RIC)

Two Modular Avionic Units (MAU) incorporating the following major subsystems and/or functions:

- Vehicle Monitoring System (VMS) dual

- Monitor Warning System (MWS) dual

- Aural Warning Generator (AWG) single

- Central Maintenance Computer (CMC) single

- Provision for CVR/FDR data interface (single)

Pilot and co-pilot headsets

Pilot and co-pilot interphone control (cyclic grip and floor switches)

Left Modular Radio Cabinet (MRC) with the following modules:

- One VHF Comm.

- One VOR/LOC/GS/MB

Right Modular Radio Cabinet (MRC) with the following modules:

- One VHF Comm.

- One VOR/LOC/GS/MB

- One ADF

- One DME

Two pedestal mounted Multifunction Control Display Units (MCDU)

Emergency Locator Transmitter (ELT)

One flight guidance controller

Pilot and co-pilot digital audio panels with remote ICS audio port for ground operation

Pilot clock (digital)

Co-pilot clock (digital)

One magnetic compass

One outside air temperature indicator

Two Master Warning Lights (MWL)

Two Master Caution Lights (MCL)

Engine 1 fire light

Engine 2 fire light

Baggage compartment smoke detector light

Dual electrical power connection for CVR/FDR (FAA requirement compliant)

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STANDARD INTERIOR

Pilot and co-pilot crashworthy seats (with inertial reels and safety belts)
Cockpit dome/storm light
Cockpit utility lights (2)
28 V DC cockpit/cabin power outlet
Cockpit panel sun-glare shields
Overhead cockpit windows sun shades
Fluorescent emergency exit signs
Cockpit map and data case in central console
Two flash lights for pilot and co-pilot
Map and data cases in pilot and co-pilot doors
ICS panel with plug for one headset (LH side)
First aid kit

AW PREMIUM EQUIPMENT

Forced fan ventilation
Bleed air heater and defroster with air noise suppressions
Cockpit air ventilation system
Baggage compartment cargo tie-down fittings
Rotor brake
1st radar altimeter RT-300
2nd radar altimeter RT-300
4-axis dual Digital Automatic Flight Control System (DAFCS) Basic incorporated into Modular Avionic Units (MAUs)
Flight Director System (dual) incorporated into the Modular Avionic Unit (MAU)
Two Honeywell GPS modules (GNSS/SBAS – Global Navigation Satellite System/Satellite Based Augmentation System) with Flight Plan Management System (FPMS)
One Mode S diversity transponder with Extended Squitter (ADSB-Out 1090 MHz) incorporated into the Modular Radio Cabinet (MRC)
Cockpit Voice Recorder & Flight Data Recorder (CVR/FDR) with Underwater Locator Beacon (ULB)

MISCELLANEOUS / GROUND EQUIPMENT

Cargo net in baggage compartment
Covers and flags: engine exhaust pipes (2), engine air inlets (2) and Pitot tubes (2),
Manuals: flight and maintenance and overhaul, illustrated parts catalogue
Tie-down assemblies for main rotor blades (5)
Tail rotor blades flapping block (1)
Main rotor blades sock pole (1)
Main landing gear wheel chocks (2)
Main rotor balance and tracking chart, tail rotor balance chart
Nose landing gear centre pin (1) and landing gear handle locking pin (1)
Tow Bar
Loose equipment bag (1)

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VIP INTERIOR

'FiveStar' interior layout - 7 seats

- 3 armchairs leather covered (with inertial reels, sliding inflight direction, reclining backrest and separate 4-point safety belts)
- Comfortable seat leather covered (with safety belts and under-seat skirts) (Qty 4)
- Multipurpose cubby with sliding doors
- Passenger cabin hinged doors Including: Extra Silent soundproofing, leather & leatherette covered liners, one (1) metal finished handle

- Electrically operated passenger steps (LH and RH) with boarding lights (in lieu of fixed steps)
- Customized VIP cabin finishing
- Decoration trim
- Passenger carpet
- P.A. (Passenger Address) system (including loudspeakers and safety belts lights)
- ICS in the cabin for all passengers
- Passenger cabin thermo acoustic double layer tinted windows
- Cabin vertical handles

- Alternate Current (AC) power outlet
- Coat hooks

'FiveStar' comfort package

- Air conditioning system
- AVCS
- Hydraulic pulsation dampeners
- MGB Premium grade
- MVA
- Nose landing gear doors

MISSION EQUIPMENT

Increased Gross Weight - 6,800 kg (14,991 lb)

Approach plates chart holders with lights for pilot and co-pilot

Pilot and co-pilot adjustable seats (in lieu of standard seats)

Heated Glass Windshield

4-axis DAFCS (Digital Automatic Flight Control System)

Enhanced with Hover Mode (HOV)

Enhanced Ground Proximity Warning System (EGPWS) provision

Enhanced Ground Proximity Warning System (EGPWS) removable

TCAS I KTA-970 Honeywell Bendix/King provision

TCAS I KTA-970 Honeywell Bendix/King removable

(*) Required for VFR Night single pilot operations

Weather Radar Primus 660 Honeywell removable

Weather Radar Primus 660 Honeywell provision

Lightning Sensor System Honeywell LSZ-860

5th display in cockpit

ADELT (2-frequency Automatic Deployable ELT w/ GPS interface)

(*) In lieu of standard ELT

Moving Map Euroavionics EuroNav V - Provision

Moving Map Euroavionics EuroNav V - Removable

Auxiliary transversal fuel tank 132 USgal - Provision

Auxiliary transversal fuel tank 132 USgal - Removable

Emergency floats provision

Emergency floats removable

Emergency Floats Kevlar covers

Engine air particle separator (EAPS) - Provision

Engine air particle separator (EAPS) - Removable

Ice detector system

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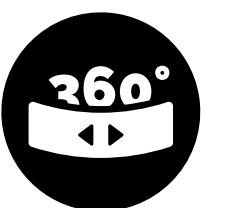
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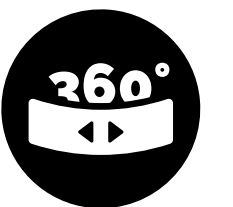
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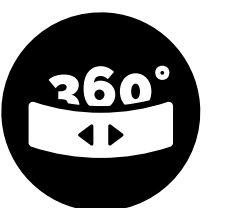
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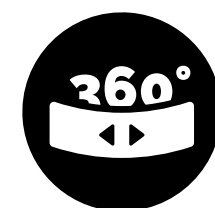
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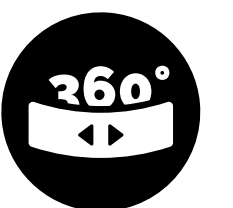
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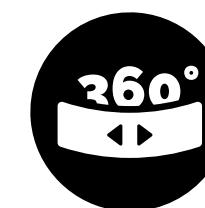
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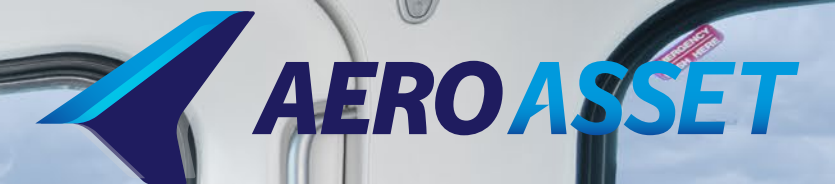


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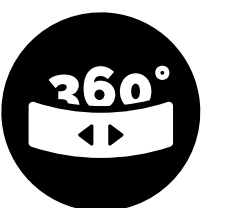
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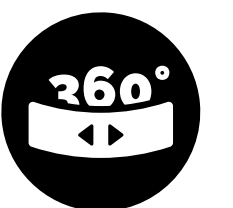
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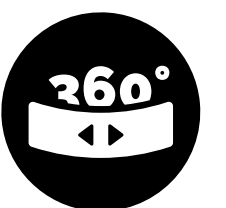
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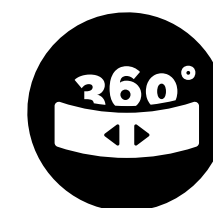
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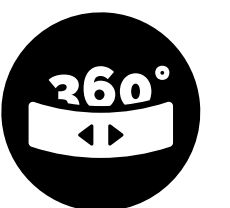
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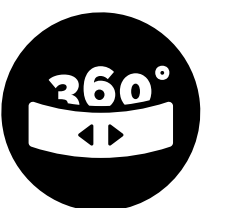
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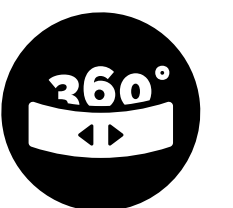
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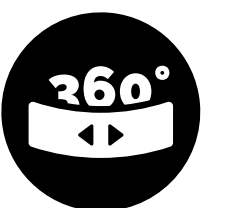
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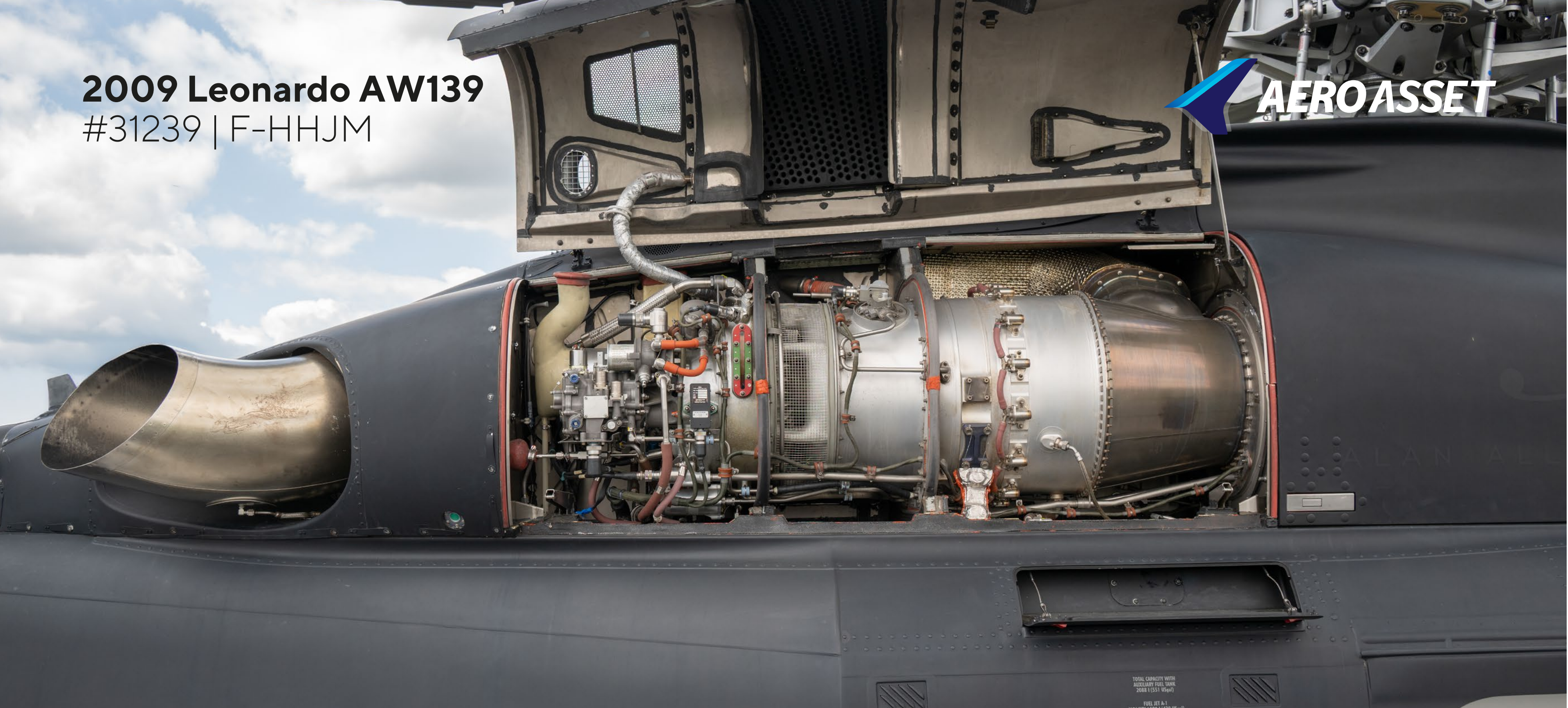


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