

Exported Listing



**Airbus AS365N3**

**EUR 2,000,000**

<b>Manufacturer</b>	Airbus
<b>Category</b>	n/a
<b>Configuration</b>	Offshore
<b>Year</b>	2009
<b>Time</b>	4325
<b>Price</b>	2000000
<b>Serial Number</b>	n/a
<b>Registration Number</b>	n/a



For inquires you can reach the seller at:

Email Address: [stephen@pacificairhub.com](mailto:stephen@pacificairhub.com) | Phone Number: [+64212803951](tel:+64212803951)

## Description

Price reduced.....contact us to discuss Oil and Gas Configuration. The cabin floor is capable of the various optional types of arrangements, and the first aid kit is EU-OPS compliant. Avionics includes air data computer Shadin ADC-2000, GPS navigation Freeflight 2101 I/O approach+ and HUMS Full (M'ARMS). Location: Ivory Coast



## Avionics

### AVIONICS

- VHF-AM ROCKWELL COLLINS (2X) KTR 908
- INTER COMMUNICATION SYSTEM TEAM TB 31 WITH 3 CONTROL BOXES 1976 (1 IN CABIN)
- PA WITH 4 LOUDSPEAKERS NAT AA20-431
- THALES AHV-16 RADIO ALTIMETER
- ATC TRANSPONDER MST67A (MODE-S)
- SSCVFDR HONEYWELL AR COMBI
- ADF HONEYWELL KDF-806
- DME HONEYWELL KDM 706A
- VOR/ILS HONEYWELL KNR-634A (2X)
- AIR DATA COMPUTER SHADIN ADC-2000
- GPS NAVIGATION FREEFLIGHT 2101 I/O APPROACH+,
- SFIM CDV 85 D3 FLIGHT DIRECTOR COUPLER
- TELEPHONICS RDR1600C WEATHER/SEARCH RADAR
- UMS
- HUMS FULL (M'ARMS)

### INSTRUMENTS

- 2 X2 PFD/ND COLLINS MFD-255
- 1 STANDBY GYRO HORIZON
- 1 STANDBY ALTIMETER
- 1 STANDBY AIRSPEED INDICATOR
- 1 DUAL TORQUE, ON PILOT'S SIDE
- 2 ENGINES EXHAUST PIPES TEMPERATURE INDICATOR
- 2 ENGINES OIL PRESSURE AND TEMP INDICATORS
- 2 FUEL PRESSURE INDICATORS

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- 1 MAIN GEARBOX OIL PRESSURE AND TEMP INDICATOR
- 2 HYDRAULIC PRESSURE GAUGES
- 1 VOLTMETER
- 1 AMMETER
- 1 DUAL FUEL CONTENTS GAUGE
- 1 FUEL CIRCUIT CONTROL AND INSPECTION PANEL
- 1 ELECTRICAL CONTROL PANEL
- 2 STOPWATCHES
- 1 TRIPLE TACHOMETER FOR ROTOR AND 1&2 FREE TURBINES RPM ON PILOT'S SIDE
- 1 ROTOR TACHOMETER ON CO-PILOT'S SIDE
- 2 NG INDICATORS (GAS GEN 1&2)
- 1 STANDBY MAGNETIC COMPASS
- 1 OUTSIDE AIR TEMPERATURE INDICATOR
- 1 WARNING PANEL
- 2 MASTER ALARM LIGHTS
- 2 MANOEUVRE LIMIT WARNING LIGHT
- 1 OVERHEAD PANEL INCLUDING ENGINE CONTROL PANEL WITH 2 FIRE WARNING LIGHTS AND 2 DUAL FIRE EXTINGUISHING CONTROLS FOR ENGINE BAYS
- 1 "L/G NOT EXTENDED" WARNING LIGHT
- SPARE SPACE FOR RADAR SCREEN AND COM/NAV EQUIPMENT
- ICING DETECTOR (ROSE MOUNT)
- 1 LANDING GEAR POSITION SELECTOR AND INDICATOR
- 1 AFCS CONTROL PANEL

## Equipment

### FLIGHT CONTROLS

- 1 MAIN ROTOR WITH 4 GLASS AND CARBON-FIBRE BLADES WITH STARFLEX HEAD FITTED WITH GUST AND DROOP STOPS, MAST FITTED WITH ROTOR RPM PHONIC WHEEL
- 1 FENESTRON TYPE TAIL ROTOR WITH COMPOSITE MATERIAL BLADES BUILT INTO THE VERTICAL FIN
- 1 FLIGHT CONTROL SYSTEM, FITTED WITH 3 DUAL-CHAMBER/DUAL BODY MAIN SERVO UNITS(CYCLIC AND COLLECTIVE PITCH CHANNELS) AND 1 DUAL-CHAMBER/DUAL BODY REAR SERVO-UNIT (ON TAIL ROTOR PITCH CONTROL CHANNEL)
- 1 SFIM PA155D DUPLEX-TYPE AUTO PILOT

### ADDITIONAL EQUIPMENT

- EMERGENCY FLOATATION WITH AFDS

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- EXTERNAL AERAZUR (SR-10) LIFE RAFT INSTALLATION, INCLUDING CABIN FOOTSTEPS
- HELICOPTER EMERGENCY EGRESS LIGHTING (HEEL)
- HR-SMITH CPI-503 ADELT
- DUKANE DK120 ULB
- AVAD RACAL V694 100FT, CH, UC WARNING
- FRONT GLASS INSTEAD OF PLEXIGLASS

#### RETROFITS

- SATCOM (COMMUNICATION AND TRACKING) (SKYROUTER)
- SSQAR INSTALLATION
- VHF/FM INSTALLATION (NPX-138)
- DF INSTALLATION (CHELTON)

#### PROVISIONS

- SAND FILTERS, FIXED PARTS
- AUXILIARY FUEL TANK IN THE HOLD, FIXED PARTS
- GOODRICH HOIST CLASS1 (272KG) FIXED PARTS
- CARGO SLING WITH DYNAMOMETER AND ELECTRICAL MIRROR (1600KG) FIXED PARTS

### Interior

#### CABIN

- 1 CABIN FLOOR CAPABLE OF THE VARIOUS OPTIONAL TYPES OF ARRANGEMENTS
- 2 REMOVABLE PILOT AND CO-PILOT HIGH BACK-REST SEATS, ADJUSTABLE IN REACH AND HEIGHT EACH FITTED WITH A SAFETY BELT AND SHOULDER 4-POINT HARNESS
- 11 STANDARD SEAT LAY-OUT
- 4 POINTS HARNESSES FOR 11 STANDARD SEAT LAYOUTS
- 2 PILOT AND CO-PILOT JETTISON ABLE DOORS, EACH FITTED WITH A BAD WEATHER WINDOW AND AN INTERNAL STORAGE UNIT. DOORS WITH AIR INLETS
- 2 HINGED JETTISON ABLE CABIN FRONT DOORS
- 2 ENLARGED AND BUBBLED REAR SLIDING DOORS WITH JETTISON ABLE WINDOW FROM INSIDE AND OUTSIDE
- 2 RETRACTABLE FOOTSTEPS
- CABIN UPHOLSTERY WITH SOUND PROOVING
- 1 INSTRUMENT PANEL, ONE CONSOLE AND ONE CEILING PANEL
- DUAL FLIGHT CONTROL
- ENGINE CONTROLS
- ROTOR BRAKE CONTROL

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- 1 HEATING/DEMISTING/VENTILATION SYSTEM
- 2 UPPER TINTED PANES
- 2 FRONT PLEXIGLASS PANES
- 2 WINDSHIELD WIPERS
- 2 ILLUMINATED CHART HOLDERS
- 2 HEADSET HOLDERS (SILEX)
- 1 PORTABLE FIRE-EXTINGUISHER
- 1 FLIGHT MANUAL
- 1 FIRST-AID KIT EU-OPS COMPLIANT

### General Characteristics

- AIRFRAME CYCLES: 9,110
- FUSELAGE COMPRISING THE CABIN AND LUGGAGE HOLD
- LONG NOSE
- LUGGAGE HOLD WITH FLOOR, TIE DOWN NET AND ACCESS DOOR
- TAIL BOOM WITH STABILIZER FITTED WITH 2X LATERAL FINS AND TERMINATED BY A SHROUDED TAIL ROTOR BUILD INTO THE VERTICAL MAIN FIN
- RETRACTABLE TRICYCLE LANDING GEAR WITH AXIALLY LOCKABLE CASTERING NOSE WHEEL UNIT, ASSISTED DIFFERENTIAL BRAKES ON PILOTS AND CO-PILOTS STATION AND PARKING BRAKE
- 3 HEATED PITOT HEADS
- BUILD IN FOOTSTEPS 2 ON EACH SIDE FOR ACCESS TO THE TRANSMISSION DECK
- ANTI-CORROSION PROTECTION
- FIXED PARTS FOR MAIN ROTOR BLADE FOLDING
- STRUCTURAL PROVISIONS FOR 1600KG CARGO SLING
- STRUCTURAL REINFORCEMENTS FOR HOIST
- JACKING, HOISTING, MOORING AND GRIPPING POINTS

### POWER PLANT

- 1 FUEL SYSTEM INCL. 5 TANKS SPLIT INTO 2 GROUPS, WITH A TOTAL USABLE CAPACITY OF 1135 LITRES, 4 IMMersed BOOSTER PUMPS, 1 TRANSFER PUMP AND AN INDICATION OF LOW LEVELS
- 2 ENGINES LUBRICATION AND OIL COOLING SYSTEMS
- 2 FIRE DETECTION AND EXTINGUISHING SYSTEMS
- 2 ANTI-ICING AIR INTAKE GRIDS
- 2 PHASE ANGLE TORQUE METER SENSORS BUILT INTO THE ENGINES
- 2 ENGINES EXHAUST PIPES

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## TRANSMISSION SYSTEM

- 1 MAIN GEARBOX
- 1 TAIL GEARBOX
- 1 MAIN GEARBOX OIL COOLING SYSTEM
- 2 ENGINE/MAIN GEARBOX COUPLING SHAFTS
- 1 TAIL ROTOR DRIVE SHAFT
- 1 ROTOR BRAKE
- 2 FREE WHEELS INTEGRAL WITH MAIN GEARBOX

## HYDRAULIC SYSTEM

- 2 INDEPENDENT HYDRAULIC SYSTEMS FEEDING THE SERVO-UNITS, LANDING GEAR ACTUATION SYSTEM AND ASSISTED BRAKES
- SELF-SEALING HYDRAULIC GROUND COUPLING
- 1 STAND-BY HYDRAULIC SYSTEM WITH ELECTRO-PUMP FOR EMERGENCY ACTUATING OF THE LANDING GEAR, HYDRAULIC ASSISTANCE FOR FLIGHT CONTROLS TESTS ON GROUND WITH ROTOR STOPPED AND PARK BRAKE PRESSURE

## ELECTRICAL INSTALLATION

- 2 STATIC INVERTERS (800VA, 115/26V, 400HZ, 1PHASE)
- 2 STARTER GENERATORS (4,8 KW, 28VDC)
- 1 43AMP-HR CADMIUM-NICKEL BATTERY WITH TEMP DETECTOR
- 1 EXTERNAL DC POWER RECEPTACLE
- 1 INSTRUMENT WHITE/BLUE LIGHTING SYSTEM
- 2 CABIN EXTENSION LIGHTS
- 1 INSTRUMENT LIGHT FOR FLIGHT IN STORMY CONDITIONS
- 1 CABIN DOME LIGHT
- 4 CABIN LIGHT STRIPS
- 1 LUGGAGE HOLD DOME-LIGHT
- 3 POSITION LIGHT
- 1 ANTI-COLLISION LIGHT
- RETRACTABLE LH LANDING LIGHT, ADJUSTABLE IN ELEVATION
- RETRACTABLE RH LANDING LIGHT, ADJUSTABLE IN ELEVATION AND IN AZIMUTH ALLOWING SEARCH LIGHT AND HOIST LIGHT
- 1 CABIN POWER OUTLET (28VDC)
- 1 EMERGENCY BATTERY FOR AUTOMATIC LIGHTING OF THE DOME LIGHTS AND SIGNS
- UNDER FUSELAGE ANTI-COLLISION STROBE LIGHT

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