

Exported Listing



**Airbus AS365N2**

**Contact Seller For Price**

<b>Manufacturer</b>	Airbus
<b>Category</b>	n/a
<b>Configuration</b>	Search and rescue
<b>Year</b>	1991
<b>Time</b>	9280
<b>Price</b>	n/a
<b>Serial Number</b>	n/a
<b>Registration Number</b>	n/a

**Description**

- CONFIGURATION: SAR WITH CVR  
INSTALLATION - GOODRICH HOIST CLASS 2  
(272KG) FIXED PARTS - CARGO SLING WITH  
DYNAMOMETER AND - ELECTRICAL MIRROR  
(1,600KG) FIXED PARTS

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## Avionics

- VHF-AM VHF-422C (2X) (8.33KHZ UPGRADED)
- INTER COMMUNICATION SYSTEM TEAM TB 31 WITH 3 CONTROL BOXES 1976 (1 IN CABIN)
- PA WITH 4 LOUDSPEAKERS TEAM BA1920
- THALES AHV-16 RADIO ALTIMETER
- ATC TRANSPONDER GARMIN GTX-330 (MODE-S)
- SAT COM LATTITUDE
- VHF/FM MARITIME NAT NPX-138
- CVR HONEYWELL 30B
- ADF KING KDF-806
- DME KING KDM-706A
- VOR/ILS KING KNR-634A
- AIR DATA COMPUTER
- GPS MARCONI CMA3012
- TELEPHONICS RDR1400 WEATHER/SEARCH RADAR
- RADAR INTERFACE IU2023
- JEPPESEN MAP SKYFORCE SKYMAP IIIC

## Equipment

### INSTRUMENTS

- 2 ALTIMETERS
- 2 AIRSPEED INDICATORS WITH DIGITAL SPEED DISPLAY
- 2 GYRO HORIZON
- 2 SFIM CG130 + 4" HSI
- 1 SELECTOR SWITCH FOR GYROCOMPASS #1 & #2
- 2 RATE-OF-CLIMB INDICATORS, WITH PROVISIONS FOR RATE-OF-CLIMB
- PRE-SET ON PILOTS' SIDE
- 1 RMI ON PILOTS' SIDE
- 1 STANDBY GYRO HORIZON
- 3 HEATED PITOT HEADS
- DUAL TORQUEMETER ON PILOT SIDE
- TAIL PIPE TEMPERATURE INDICATORS
- 2 ENGINE OIL PRESSURE AND TEMPERATURE INDICATORS
- 2 FUEL PRESSURE INDICATORS

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- 1 MAIN GEARBOX OIL PRESSURE AND TEMP INDICATOR
  - 2 HYDRAULIC PRESSURE GAUGES
  - 1 VOLTMETER
  - 1 AMMETER
  - 1 DUAL FUEL CONTENTS GAUGE
  - 1 FUEL CIRCUIT CONTROL AND INSPECTION PANEL
  - 1 ELECTRICAL CONTROL PANEL
  - 1 LANDING GEAR POSITION SELECTOR AND INDICATOR
  - 2 STOPWATCHES
  - 1 TRIPLE TACHOMETER FOR ROTOR AND 1&2 FREE TURBINES RPM ON PILOT'S SIDE
  - 1 ROTOR TACHOMETER ON CO-PILOT'S SIDE
  - 2 NG INDICATORS (GAS GEN 1&2)
  - 1 STANDBY MAGNETIC COMPASS
  - 1 OUTSIDE AIR TEMPERATURE INDICATOR
  - 1 WARNING PANEL
  - 2 MASTER ALARM LIGHTS
  - 2 MANOEUVRE LIMIT WARNING LIGHT
  - 2 FIRE WARNING LIGHTS AND 2 DUAL FIRE EXTINGUISHING CONTROLS FOR ENGINE BAYS
  - 1 "L/G NOT EXTENDED" WARNING LIGHT
  - SPARE SPACE FOR RADAR SCREEN AND COM/NAV EQUIPMENT
  - ICING DETECTOR (ROSE MOUNT)
- ADDITIONAL EQUIPMENT
- DIRECTION FINDER CHELTON DF7 SYSTEM
  - DOPPLER INSTALLATION, RACAL D91
  - HELICOPTER EMERGENCY EGRESS LIGHTING (HEEL)
  - HR-SMITH CPI-503 ADELTA
  - DUKANE DK120 ULB
  - AVAD RACAL V694 100FT, CH, UC WARNING
  - SAND FILTERS, FIXED PARTS
  - GOODRICH HOIST CLASS2 (272KG) FIXED PARTS
  - SPECTROLAB SX-5 SEARCH LIGHT
  - CARGO SLING WITH DYNAMOMETER AND ELECTRICAL MIRROR (1600KG) FIXED PARTS
  - RN NAV2 + DOUPLER D91
  - DE-ICED WINDSHIELD

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- FLIR EO/IR HARD POINTS
- HEEL SYSTEM
- UHF RADIO INSTALLATION
- AIS TRANSPONDER INSTALLATION
- SAR CABIN LAYOUT / CVR INSTALLATION

## Interior

- 1 CABIN FLOOR CAPABLE OF THE VARIOUS OPTIONAL TYPES OF ARRANGEMENTS
- 2 REMOVABLE PILOT AND CO-PILOT HIGH BACK-REST SEATS, ADJUSTABLE IN REACH AND HEIGHT EACH FITTED WITH A SAFETY BELT AND SHOULDER 4-POINT HARNESS SAR LAY-OUT
- 2 PILOT AND CO-PILOT JETTISONABLE DOORS, EACH FITTED WITH A BAD WEATHER WINDOW AND AN INTERNAL STORAGE UNIT.
- 2 HINGED JETTISONABLE CABIN FRONT DOORS
- 2 ENLARGED AND BUBBLED REAR SLIDING DOORS WITH JETTISONABLE WINDOW FROM INSIDE AND OUTSIDE
- 4 RETRACTABLE FOOTSTEPS
- 2 UPPER TINTED PANES
- CABIN UPHOLSTERY WITH SOUND PROVING
- 1 INSTRUMENT PANEL, ONE CONSOLE AND ONE CEILING PANEL
- DUAL FLIGHT CONTROL
- ENGINE CONTROLS
- ROTOR BRAKE CONTROL
- 1 HEATING/DEMISTING/VENTILATION SYSTEM
- 2 WINDSHIELD WIPERS
- 2 ILLUMINATED CHART HOLDERS
- 2 HEADSET HOLDERS (SILEX)
- 1 PORTABLE FIRE-EXTINGUISHER

## General Characteristics

### FLIGHT CONTROLS

- 1 MAIN ROTOR WITH 4 GLASS AND CARBON-FIBRE BLADES WITH STARFLEX HEAD FITTED WITH GUST AND DROOP STOPS, MAST FITTED WITH ROTOR RPM PHONIC WHEEL
- 1 TAIL ROTOR WITH 11 COMPOSITE MATERIAL BLADES BUILT INTO THE VERTICAL FIN
- 1 FLYING CONTROL SYSTEM, FITTED WITH 3 DUAL-CHAMBER/DUAL BODY MAIN SERVO UNITS (CYCLIC AND COLLECTIVE PITCH CHANNELS) AND 1 DUAL-CHAMBER/DUAL BODY REAR SERVO-UNIT (ON TAIL ROTOR PITCH)

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#### CONTROL CHANNEL)

- 1 SFIM CDV 155 FLIGHT DIRECTOR COUPLER (4-AXIS) (WITH 2ND RADAR ALTIMETER)

#### ELECTRICAL INSTALLATION

- 2 250VA, 115/26V, 400 HZ 1-PHASE STATIC INVERTERS
- 2 STARTER GENERATORS (4,8 KW, 28VDC)
- 1 43AMP-HR CADMIUM-NICKEL BATTERY WITH TEMP DETECTOR
- 1 EXTERNAL DC POWER RECEPTACLE
- 1 INSTRUMENT WHITE/BLUE LIGHTING SYSTEM
- 2 CABIN EXTENSION LIGHTS
- 1 INSTRUMENT LIGHT FOR FLIGHT IN STORMY CONDITIONS
- 1 CABIN DOME LIGHT
- 4 CABIN LIGHT STRIPS
- 10KVA ALTERNATOR
- 1 LUGGAGE HOLD DOME-LIGHT
- 3 POSITION LIGHT
- 1 ANTI-COLLISION LIGHT
- RETRACTABLE LH LANDING LIGHT, ADJUSTABLE IN ELEVATION
- RETRACTABLE RH LANDING LIGHT, ADJUSTABLE IN ELEVATION AND IN AZIMUTH ALLOWING SEARCH LIGHT AND HOIST LIGHT
- 1 CABIN POWER OUTLET (28VDC)
- 1 EMERGENCY BATTERY FOR AUTOMATIC LIGHTING OF THE DOME LIGHTS AND SIGNS
- 1 ROTOR TACHOMETER ON CO-PILOT'S SIDE UNDER FUSELAGE ANTI-COLLISION STROBE LIGHT

#### TRANSMISSION

- 1 MAIN GEARBOX, ANTI-VIBRATION MOUNTED, WITH OIL SIGHT GAUGE, MAGNETIC PLUG, OIL PRESSURE AND TEMPERATURE PICK-UP, LUBRICATION SYSTEM, THERMAL SWITCH, ROTOR TACHOMETER DRIVE AND HOLES FOR

#### ENDOSCOPE AND OIL SAMPLING

- 1 TAIL GEARBOX WITH OIL SIGHT GAUGE AND MAGNETIC PLUG
- 1 MAIN GEARBOX OIL COOLING SYSTEM
- 2 ENGINE/MAIN GEARBOX COUPLING SHAFTS
- 1 ROTOR BRAKE
- 2 FREE WHEELS INTEGRAL WITH MAIN GEARBOX

#### HYDRAULICS

- 2 INDEPENDENT HYDRAULIC SYSTEMS FEEDING THE SERVO-UNITS, LANDING GEAR ACTUATION SYSTEM AND ASSISTED BRAKES

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- SELF-SEALING HYDRAULIC GROUND COUPLING
- 1 STAND-BY HYDRAULIC SYSTEM WITH ELECTRO-PUMP FOR EMERGENCY ACTUATING OF THE LANDING GEAR, HYDRAULIC ASSISTANCE FOR FLIGHT CONTROLS TESTS ON GROUND WITH ROTOR STOPPED AND PARK BRAKE

#### PRESSURE

#### GENERAL

- FUSELAGE COMPRISING THE CABIN AND LUGGAGE HOLD
- LUGGAGE HOLDS WITH FLOOR, TIE DOWN NET AND ACCESS DOOR
- TAIL BOOM WITH STABILIZER FITTED WITH 2X LATERAL FINS AND TERMINATED BY A SHROUDED TAILROTOR BUILT INTO THE VERTICAL MAIN FIN
- RETRACTABLE TRICYCLE LANDING GEAR WITH AXIALLY LOCKABLE CASTERING NOSE WHEEL UNIT, ASSISTED DIFFERENTIAL BRAKES ON PILOTS AND CO-PILOTS
- STATION AND PARKING BRAKE
- BUILD IN FOOTSTEPS 2 ON EACH SIDE FOR ACCESS TO THE TRANSMISSION DECK
- FIXED PARTS FOR MAIN ROTOR BLADE FOLDING
- STRUCTURAL PROVISIONS FOR 1600KG CARGO SLING
- STRUCTURAL REINFORCEMENTS FOR HOIST
- JACKING, HOISTING, MOORING AND GRIPPING POINTS
- 1 FUEL SYSTEM INCLUDING 5 TANKS SPLIT INTO 2 GROUPS, WITH A TOTAL USABLE CAPACITY OF 1135 LITRES, 4 IMMERSSED BOOSTER PUMPS, 1 TRANSFER - PUMP AND AN INDICATION OF LOW LEVELS.
- 1 HYDRAULIC BLEED CONTROL FOR THE WHOLE FUEL SYSTEM
- 2 ENGINES LUBRICATION AND OIL COOLING SYSTEMS
- 2 FIRE DETECTION AND EXTINGUISHING SYSTEMS
- 2 ANTI-ICING AIR INTAKE GRIDS
- 2 HYDRAULIC SENSORS FOR TORQUE METER BUILT INTO THE ENGINES
- 2 ENGINES EXHAUST PIPES
- ENGINE FLUSHING DEVICE

#### Remarks

- AIRFRAME CYCLES: 15,675

*The descriptions and/or specifications provided are for introductory information only and do not imply any warranties or representations. The Purchaser is responsible for verification of specifications upon inspection. The Helicopter is offered subject to prior sale, lease or withdrawal from the market without prior notice.*

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